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American Champion Aircraft Corp. Rochester, Wisconsin

FAA Approved

Airplane Flight Manual

Citabria Aurora Model 7ECA

This manual is eligible only for 7ECA aircraft, serial numbers 1355-95 and up.

Registration Number: N608AC

Serial Number: 1356-95

This manual is part of the required equipment and must remain in the airplane at all times.

Approved: Approved

Royce Prather

Manager, Chicago Aircraft Certification Office

Date: 11/01/00

Revisions:

Letter	Ву	Date	Revisions	Pages Affected
A	JKM	1/4/96	Corrected Typos, Added Revision Table, Rearranged Sections	All
В	ЈКМ	10-13- 2000	Revised to include new 1750 lbs. gross weight changes and limitations	All

Flight Operations

All 7ECA models are approved in the Normal and Acrobatic Category. Day or night flight in VFR conditions only is approved providing the aircraft is equipped with the required equipment and is in operating condition as specified under Part 91 of the Federal Air Regulations. Flight into known icing condition is prohibited

Airspeed Limitations	CAS (mph)
Never Exceed (red line)	162
Caution Range (yellow arc)	
Maximum Structural Cruise	
Normal Operating Range (green arc)	50 - 120
Maneuvering @ 1750	
Stall Speed	
Best Rate of Climb Speed	

CAS - Calibrated airspeed is indicated airspeed corrected for installation and instrument error

Weight and Balance Limits

Consult aircraft records for weight and balance data for this aircraft.

Aircraft Weight and Center of Gravity Range (inches)

Weight	Normal Category	Acrobatic Category
1750 lbs. maximum	14.2 - 19.2	14.2 - 16.3
1325 lbs or less	105-192	105-163

All measurements are aft of the datum line which is the wing leading edge. Center of gravity limits have straight line variation between these points.

Flight Load Factors (1750 lbs. normal or acrobatic category)

Positive Load Factor+	5.0	g's maximum
Negative Load Factor	2.0	g's maximum

Do not exceed 135 mph IAS during negative g-load condition. Do not perform acrobatics in turbulent air.

11/01/00

Maneuvers

Landplane Approved For Only The Following Acrobatic Maneuvers

	Entry Speed IAS		
Maneuver	MPH	Knots	
Chandelle, Lazy Eight	120	104	
Barrel or Slow Roll	120	104	
Immelmann	145	126	
Loop or Clover Leaf	140	122	
Split S	80	70	
Snap Roll	85	74	
Hammerhead Turn	140	122	
Cuban Eight	145	126	
Spin	Slow Dec	celeration	

Do not exceed +5.0 or -2.0 g-load factor. Do not exceed 135 mph IAS during negative g-load condition. Do not perform acrobatics in turbulent air. Caution: negative g flight will cause loss of oil and oil pressure. Use of flaps during acrobatics is prohibited. To recover from spin use full opposite rudder and neutralize elevator.

Powerplant Limitations

Engine, Lycoming	O-235-K2C	
Rated Horsepower (HP / RPM)	118 / 2800 115 / 2700	Takeoff Max. Continuous
Fuel, Aviation Grade, Min. Octane	91 / 96	
Tachometer Normal Range (green arc - RPM) Takeoff (5 min) (yellow arc - RPM) Maximum (red line - RPM)	1800 - 2700 2700 - 2800 2800	
Cylinder Head Temperature Maximum (red line - °F)	500	

11/01/00

Powerplant Limitations (Continued)

Oil Temperatures

Normal Range (green arc - °F)	100 - 245
Maximum (red line - °F)	245

Oil Pressure

Normal Range (green arc - psi)	60 - 100
Caution Range (yellow arc - psi)	25 - 60
Maximum (red line - psi)	100
Minimum (red line - psi)	25

Other Operating Information

Occupy front seat when flying solo.

Fuel 35 gal. useable, fuel 'on' when handle is down.

Maximum baggage - 100 lbs. Do not carry baggage during aerobatics.

Folding rear seat, seat back restrainer cable must be connected before flight unless rear control stick is removed.

Turn off strobe lights when taxiing in vicinity of other aircraft or during flight through clouds, fog, or haze. Standard position lights to be on for all night operations.

Alternator output circuit breaker located on cross member behind instrument panel.

Emergency door hinge release - pull pin, pull lever

Lest window may be used as an alternate emergency exit - force window outward past forward stop mechanism.

Do not open forward left side window above 130 mph, or above 90 mph if the window is full opening.

Service battery every 50 hours of normal flight. Service every 10 hours of acrobatic flight

11/01/00

Noise Characteristics

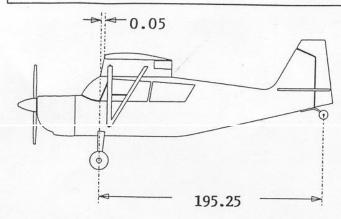
The noise level for this airplane measured in accordance with FAR 36, Appendix G is 67.0 dB(A) at full throttle.

The noise level for this airplane measured in accordance with ICAO, Annex 16, Chapter 10, Appendix 6 is 71.24 dB(A) at full throttle.

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

AMERICAN CHAMPION AIRCRAFT CORPORATION ROCHESTER, WISCONSIN 53167

WEIGHT AND BALANCE WORKSHEET



Aircraft Leveling Means: Drop plumb line from wing leading edge so that it is 10.18 inches forward of front face fuselage wing strut fitting.

Weight Actual 1143.2

. 1.	Left Wheel Weight:539.6	lbs.	Right Wheel Weight: 535.6	lbs.
2.	Tail Wheel Weight68			lbs.
3.	Total Aircraft Weight (Full Oil):			
	(<u>195.25</u> in.) X (<u>68</u>	_lbs.)		
	C.G. =	(0.05 in.) = 11.564	in Aft L.E.
	(1143.2 lbs.)			

	WEIGHT (lbs.)	ARM (in.)	MOMENT (in. lbs.)
Aircraft Weight With Full Oil	1143.2	11.564	13219.965
Subtract Drainable Oil	-11.25	-36	+405
Add 1 Gallon Useable Fuel	+6	+15	+90
TOTALS	1137.95	xxxxxxxx	13714.965

Aircraft Empty C.G. 12.05 in. aft datum

Useful Load 506.8 lbs.

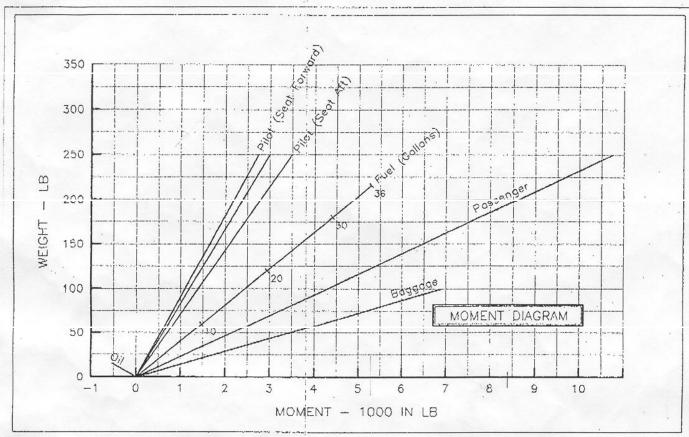
Datum: Wing Leading Edge

CENTER OF GRAVITY LIMITS: (+ 14.2 in.) To (+19.2 in.) At 1650 lbs. NORMAL CATAGORY (+ 10.5 in.) To (+19.2 in.) At 1325 lbs or less ACROBATIC CATAGORY (+ 14.2 in.) To (+17.3 in.) At 1650 lbs. or less (+ 10.5 in.) To (+16.3 in.) At 1325 lbs or less Straight line variation between points given

FORM ACAC-78A

LOADING SCHEDULE 7ECA

REVISEO: 1--03-01



Add weights and moments of items in MOMENT DIAGRAM to airplane empty weight and moment. (negative oil moment) Locate intersection of total weight and moment on AJRPLANE LOADING ENVELOPE. Any point falling within the envelope meets all balance requirements.

